

## REMARKS

By the present amendment, claims 1, 2, 3 and 5 are pending in the application. Claims 1 and 2 are independent claims.

### Claim Amendments

The amendments to claims 1 and 2 directed to the “pins” are supported in the specification, e.g., at page 7, lines 4 to 6.

### §103

Claims 1 to 5 were rejected under 35 U.S.C. §103(a) as being unpatentable over U.S. Patent No. 6,338,765 to Statnikov (the “‘765 patent”).

This rejection, as applied to the amended claims, is respectfully traversed.

### Patentability

#### The depth of the structure improved layer.

As pointed out by the Office Action, the ‘765 patent clearly mentions that microstructure can be improved by means of UIT (ultrasonic impact technology, Col. 5, lines 24-25), such as, column 5, lines 54-55 “to improve the grain structure” and column 6, lines 62-64 “modified .... material structure”. However, the ‘765 patent only describes that “white layers are formed .... typically one micron thick” (col. 7, lines 5 to 7). There is no description about a more concrete depth.

In addition, the ‘765 patent does not disclose or suggest the characteristic feature of the present invention as to average of longitudinal axis of crystal grains at a depth of at least 2 mm from the surface of the steel plate in the microstructure adjacent to a fusion line of a weld metal to equalize the grain diameter of the HAZ to a base steel plate.

Considering a description of white layers, a structure is only improved at most at the surface layers in the ‘765 patent. A person skilled in the art could not easily conceive the characteristic feature of the present invention of the average of longitudinal

axis of crystal grains at a depth of at least 2 mm from the surface of the steel plate in the microstructure adjacent to a fusion line of a weld metal to equalize the grain diameter of the HAZ to a base steel plate, and the crystal grain size of the steel plate matrix before the welding at a depth of  $\frac{1}{4}$  of the thickness from the surface of the steel plate, based on the teachings of the '765 patent.

We cannot analyze a detail UIT process described in the '765 patent. However, the attached technical publication "IIW/IIS Doc.XIII-1757-1999" Guideline For Application Of Ultrasonic Impact Treatment Improving Fatigue Life Of Welded Structure" written by S. Statnikov (same as the inventor of the '765 patent) describes a detail UIT process on page 7. Comparison with this UIT process, although the indenter diameter of this UIT process is 2-5 mm, the indenter diameter of the present invention is 5-30 mm. It is clearly seen that the indenter diameter of the present invention is quite different from that of the UIT process of this publication which relates to the '765 patent. This difference causes an increased thicker structure and improved layer from the surface of the steel plate in the present invention as compared to the '765 patent.

Means of toughness improvement.

The Office Action asserted that it is easily conceived that toughness is improved by a removal of voids and repair of cracked structure, column 10, lines 20-40. It is well known that toughness is improved by a reduction of voids cracks. However, the present invention does not rely on this teaching.

The present invention solves a problem of toughness improvement by means of subjecting a surface of a heat affected zone formed by a last pass of a multi-layer welded joint of a steel plate to impacts by an ultrasonic vibration tool using one or more pins having a diameter of 5 to 30 mm with an oscillating amplitude of between 20 to 60  $\mu\text{m}$  to

thereby make an average of longitudinal axis of crystal grains at a depth of at least 2 mm from the surface of the steel plate in the microstructure adjacent to a fusion line (FL) of a weld metal and a steel plate matrix in said heat affected zone formed by the last pass equivalent to the crystal grain size of the steel plate matrix before the welding at a depth of  $\frac{1}{4}$  of the thickness from the surface of the steel plate. This inventive idea cannot be conceived by a person skilled in the art based on the '765 patent and such a technical publication.

Further, it is submitted that it is difficult to conceive that toughness improvement can be seen by a bulk impact test using 10 mm x 10 mm test piece for a Charpy impact test based on the teaching of a white layer in the '765 patent. In addition, it is well known that steel becomes brittle and lowers toughness when plastic strain is applied. Although the UIT process is useful to apply large plastic strain to the steel, it cannot be easily applied to a steel plate for improving toughness.

Fatigue strength and toughness.

It is well known that there is no relationship between toughness and fatigue strength. According to the attached standard "Fatigue Design Of Welded Joints And Components" XIII-1539-96/XV-845-96, The International Institute of Welding, fatigue strength and fatigue life only depend on the applied force and the shape of weld joint, and does not depend on toughness level of the steel plate or weld joint. In other words, it is understood that when the shape of weld joint (if the shape of weld joint is the same, stress concentration coefficient and weld retained stress are the same), the same level of fatigue strength and same fatigue life are achieved.

It is therefore submitted that amended independent claims 1 and 2, and claims 3 and 4 dependent thereon, are patentable over U.S. Patent No. 6,338,765 to Statnikov.

**CONCLUSION**

It is submitted that in view of the present amendment and foregoing remarks, the application is now in condition for allowance. It is therefore respectfully requested that the application, as amended, be allowed and passed for issue.

Respectfully submitted,

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**IIW/IIS - DOCUMENT XIII - 1757 - 99**

**GUIDE FOR APPLICATION OF ULTRASONIC IMPACT TREATMENT  
IMPROVING FATIGUE LIFE OF WELDED STRUCTURES**

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**Abstract**

In recent years, in the context of fatigue life improvement methods, Ultrasonic Impact Treatment (UIT) has attracted a particular attention.

Independent expert assessments of this method provide indications of its effectiveness, workability, compatibility with welding fabrication and repair processes, controllability, simplicity in use and in quality control, high stability and reproducibility of results.

The last feature (reproducibility) is largely attributable to the proper preliminary tool and treatment parameters selection for specific materials, welded joint types and service conditions.

Practical use of this method in the fabrication or repair of welded metal structures requires the establishment of guidelines for proper selection of UIT parameters. This selection is made based on desired results of treatment, types of treated materials and welded joints.

The paper is submitted for consideration to Commission XIII  
of the International Institute of Welding  
1999 Lisbon, Portugal

## 1. Introduction

Ultrasonic Impact Treatment (UIT) was developed to improve quality, carrying capacity and life of welded joints in metal structures with long service life [1]. Positive results of fatigue testing [2, 3] and UIT applications [4] are well known. IIW UIT Specification was issued in 1996 [5]. These studies and technical applications were preceded by careful testing of UIT parameters for specific welded joints and materials at the laboratory of NSTC in Severodvinsk, Russia and at the E.O. Paton Electric Welding Institute in Kiev, Ukraine. In the past two years UIT has been further developed by Applied Ultrasonics in Birmingham, Alabama, USA in cooperation with NSTC. Additional data on UIT efficiency have been obtained [6, 7]. Algorithms of UIT operational procedures have been developed for fabrication, maintenance, and repair of welded joints with considerations for metal properties and welded joint types. The design and technical parameters of UIT equipment are improved and new series of equipment is fabricated. This document is intended for specialists in fabrication of welded structures and in fatigue life improvement. It describes the current stage of UIT method development and provides basic information to assist in selection of UIT operational procedures with the aim to increase the fatigue strength of welded structures.

## 2. UIT Mechanism and its Effect on Welded Joint Materials

The UIT mechanism of operation is represented in Fig. 1 and Fig. 2. The following sequential effects are taking place during UIT operations:

- forced oscillations 1 of ultrasonics transducer I;
- transfer of ultrasonic oscillations 1 to a replaceable (removable) concentrator of oscillating velocity (waveguide) II;
- impact of the output end of the waveguide II upon the indenter III;
- impact of the indenter III upon the treated surface of the workpiece IV;
- transformation of the oscillation 1 into force impulses 2 at the output end of the waveguide II and surface IV during the impact of the indenter III on the surface of the workpiece IV.

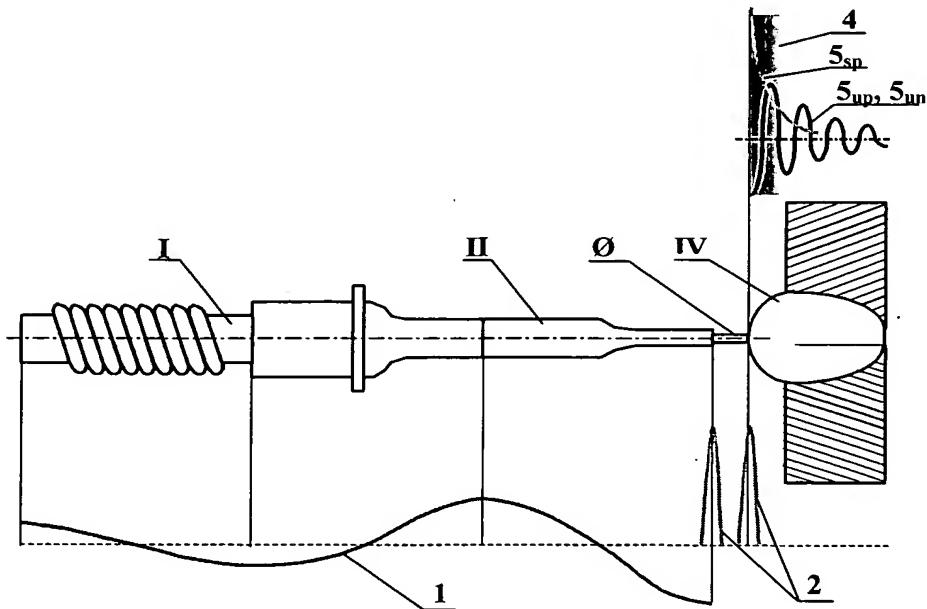
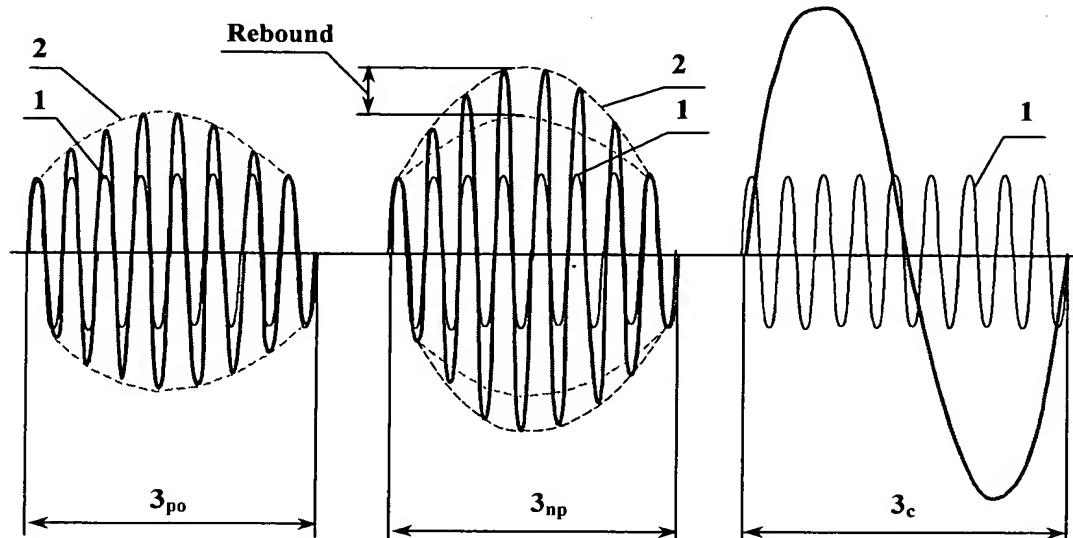


Fig. 1 UIT Mechanism

Impacts of the indenter **III** upon the workpiece surface **IV** are accompanied by one of the following interactions between the indenter **III** and the workpiece **IV** during the impact:

- ultrasonic periodic forced oscillations  $3_{po}$  of the indenter **III** in the workpiece material **IV** with continuous contact between indenter and treated surface (ultrasonic periodic impact);
- ultrasonic non-periodic forced oscillations  $3_{np}$  of the indenter **III** with indenter **III** rebounding off the workpiece surface **IV** (ultrasonic non-periodic impact);
- single contacts  $3_c$  of the indenter **III** with its rebound off the workpiece **IV**.



**Fig. 2 Interactions between indenter III and workpiece IV during impact**

Force pulses 2 initiated by the waveguide II impact upon the indenter **III**, and then by the indenter **III** upon the workpiece **IV** result in plastic deformation 4 at and under the surface of the workpiece **IV**. These impulses also set off forced oscillations 3 of the indenter **III** in the workpiece material **IV**, as well as the indenter **III** rebound off the workpiece surface **IV**.

The indenter **III** oscillations  $3_{po}$  in the workpiece material **IV** during the impact of the indenter **III** upon the workpiece **IV** and plastic deformation 4 of the workpiece material **IV** excite ultrasonic periodic stress waves  $5_{up}$  in the workpiece **IV**.

Oscillations  $3_{np}$  of the indenter **III** with a rebound off the workpiece surface **IV** during the indenter **III** impact upon the workpiece **IV** and plastic deformation 4 of the workpiece material **IV** set off propagation of ultrasonic (non-periodic) stress pulses  $5_{un}$  in the workpiece.

Single contacts  $3_c$  of the indenter **III** with its rebound off the workpiece **IV** cause propagation of single stress pulses  $5_{sp}$  in the workpiece.

The energy of force pulses 2 and oscillations 3 during indenter **III** impact on the workpiece **IV** is sequentially utilized for plastic deformation 4 of the workpiece surface **IV**, saturation of the near-surface layer of the workpiece material **IV** with plastic deformations, oscillations and pulsed deflection of this layer during the impact, and creation of ultrasonic stress waves and force pulses 5 in the volume of treated material.

Plastic deformation 4 of the treated material induces compressive stresses at the near-surface layer of the workpiece **IV**. Consequently, ultrasonic stress waves and force pulses 5 relax residual (welding) stresses in the depth of treated materials.

Indenter **III** impacts upon the workpiece surface **IV**, indenter **III** oscillations 3 during these impacts and rebounds off the surface can have features of random and controlled events. The nature of these events depends on the algorithm of oscillating system excitation **I-II-III-IV** and impact control algorithm in this system.

### 3. UIT Basic Parameters

UIT is inducing compressive stresses  $\sigma_y$  in the near-surface layer of the workpiece, is redistributing and relaxing (reducing) residual stresses in the welded joint and in whole structure.

Each of these effects defines new deflected mode  $dm$  of the welded joint material and welded structure treated by UIT, and is a function of the force pulse ( $mv$ ) at the treated surface and impact energy ( $mv^2$ ) upon this surface during UIT.

$$dm = F(mv, mv^2),$$

where:

$m$  is the equivalent mass of the oscillating system **I-II-III-IV** reduced to the output end of the waveguide **II** when applying UIT to induce compressive stresses in the surface layer of the workpiece **IV** ( $m \sigma_y$ ) and reduced to the surface of the workpiece **IV** in the treatment area when using UIT to redistribute or relax (reduce) residual welding stresses in the workpiece **IV** material depth ( $m_n$ ). Accordingly,

$v$  is the oscillating velocity at the output end of the waveguide **II** when applying UIT to induce compressive stresses in the surface layer of the workpiece **IV** ( $v \sigma_y$ ) and at the indenter **III** when applying UIT to redistribute or relax (reduce) residual welding stresses in the workpiece material **IV** depth ( $v_n$ ) respectively. Therefore:

$$m \sigma_y = I-II^* + III^* III, \quad (1)$$

where:

**I-II** is the full mass of the oscillating system **I-II**;

**III** is the functional factor of displacement distribution in the oscillating system **I-II**;

**III** is the full mass of the indenter **III**;

**III** is the functional factor of displacement distribution in the indenter **III**.

$$m_n = I-II^* + III^* III + IV^* IV, \quad (2)$$

where:

**IV** is the full mass of the workpiece **IV** as defined by its volume which is included in the oscillating system **I-II-III-IV**;

$M_{IV}$  is the mass of the volume of material within work piece 4 which forms part of the oscillating system **I-II-III-IV**;

**IV** is the functional factor of displacement distribution (fluctuating ultrasonic and pulse stress) in the volume **IV**.

In the calculation of the equivalent mass  $m$ , it is assumed that in the oscillating system **I-II-III-IV** and its elements with distributed parameters the instantaneous displacement:

for subsystem **I-II-III**:

$$= \pm \sigma_0 \sin(\omega x), \quad (3)$$

for workpiece **IV** material:

$$= \pm \sigma_0 e^{-\beta x} \sin(\omega x), \quad (3)$$

where:

$\sigma_0$  is the maximum displacement amplitude at the output end of the waveguide **II**, indenter **III**

or in the workpiece material **IV** respectively;

$\varnothing$  is the oscillation frequency of the waveguide **II** or indenter **III** respectively;

$\bullet = \frac{\omega}{c}$  is the wavenumber of the waveguide **II** or indenter **III** material respectively.

$c$  is the sonic velocity in the material of the waveguide **II**, indenter **III** or workpiece **IV** respectively;

$x$  is the linear coordinate of displacement;

$\beta$  is the loss factor in the material of a given welded joint.

Hence when inducing compressive stress in the material **IV**, the instantaneous oscillating velocity in the system **I-II-III** is defined as follows:

$$v_{\sigma_y} = \pm 0 \cdot \varnothing_{\sigma_y} \cdot \cos(\varnothing_{\sigma_y} \cdot x), \quad (4)$$

for workpiece material **IV**:

$$v_{\sigma_y} = \pm 0 \cdot \varnothing_{\sigma_y} \cdot e^{-\beta x} \cdot \cos(\varnothing_{\sigma_y} \cdot x), \quad (4)$$

where:

$\varnothing_{\sigma_y}$  is the oscillation frequency of the indenter **III**.

With redistributing and relaxation of residual stresses in the material **IV**, the instantaneous oscillating velocity in the system **I-II-III** is found from:

$$v_{rr} = \pm 0 \cdot \varnothing_{rr} \cdot \cos(\varnothing_{rr} \cdot x), \quad (5)$$

for workpiece material **IV**:

$$v_{rr} = 0 \cdot \varnothing_{rr} \cdot e^{-\beta x} \cdot \cos(\varnothing_{rr} \cdot x), \quad (5)$$

where  $\varnothing_{rr}$  is the oscillation frequency at the output end of the waveguide **II**.

Stress distribution in the workpiece material **IV** in depth  $x$  and at the surface is defined as:

$$\sigma_x = \sigma_s \cdot e^{\beta x}, \quad (6)$$

where  $\sigma_s$  is the surface stress.

Given depth of relaxation  $h_{rr}$  and associated minimum fluctuating (ultrasonic) stress level in the workpiece **IV**  $\sigma_{min_{rr}} = (0,15 \ 0,2) \sigma_y$ , the ultrasonic fluctuating stress at the workpiece surface **IV** can be defined as:

$$\sigma_{srr} = \frac{\sigma_{min_{rr}}}{e^{-\beta h_{rr}}}, \quad (7)$$

where  $h_{rr}$  is the preset depth of residual stress relaxation.

If it is assumed that at a given depth  $h_{\sigma_y}$  of plastic deformation the residual stress is equal to  $\sigma_y$ , then at the surface of the workpiece **IV** this stress is found from:

$$\sigma_{s \sigma_y} = \frac{\sigma_y}{e^{-\beta h_{\sigma_y}} \sigma_y}, \quad (8)$$

where  $h_{\sigma_y}$  is the given depth of the plastic deformation.

From both (7) and (8) relations, design and controlled parameters of UIT are determined.

Therefore, **initial parameters** to define UIT conditions are:

- welded joint type and geometry;
- stress concentration factor of a welded joint;
- mechanical properties of the welded joint material (yield strength  $\sigma_y$ , ultimate strength  $\sigma$ , sonic velocity  $c$  and loss factor  $\beta$ );
- preset fatigue characteristics of a given welded joint (fatigue limit  $\sigma_R$  and life  $N$  of a welded joint);
- oscillating system **I-II** behavior.

**Preset UIT parameters:**

- plastic deformation depth  $h_{\sigma_y}$ ;
- depth of stress redistribution and relaxation  $h_{tr}$ ;
- weld toe geometry after UIT.

**Design and controlled UIT parameters:**

- mass  $M_{III}$  and dimensions of the indenter **III**;
- force pulse (impulse)  $mv$ ;
- impact energy  $mv^2$ ;
- oscillating frequency  $\varnothing_{tr}$  of the waveguide end **II**;
- impact frequency  $\varnothing_{\sigma_y}$  of the indenter **III**;
- oscillating amplitude  $\varrho$  of the waveguide end **II**;
- radius  $R$  of the indenter contact surface **III**.

#### 4. UIT equipment

UIT of welded joints is performed with equipment consisting of an ultrasonic tool with operating frequency of 27, 36, 44 kHz and the associated ultrasonic generator (Fig.3).

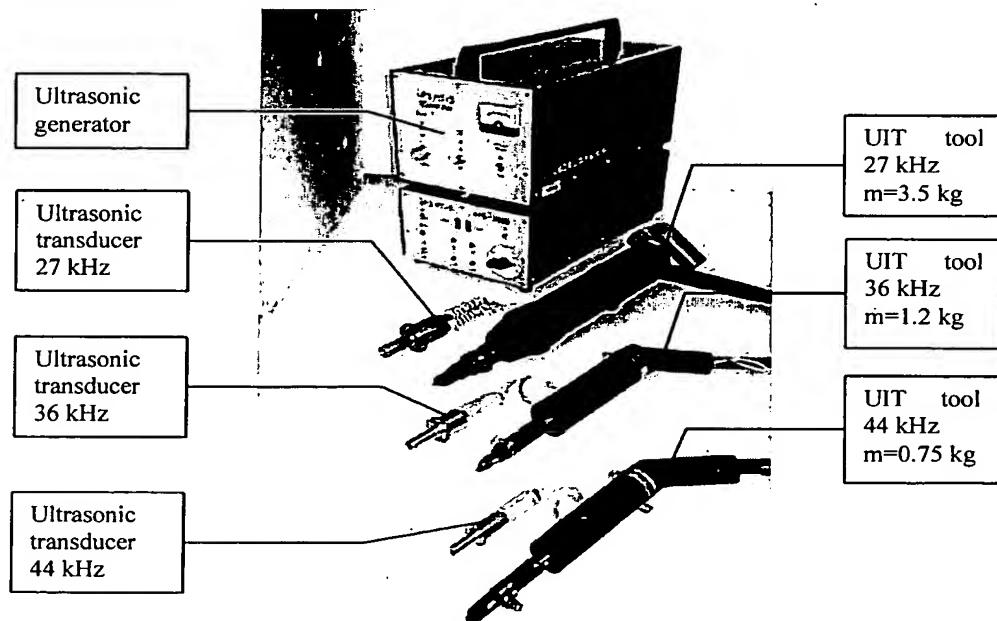


Fig. 3 UIT Equipment

### Specifications of Ultrasonic Generator UIP MSP-5

Output power	Output power adjustment	600 1800 VA
Supply main voltage		stepped (4 ranges, 6 steps in each)
Supply main frequency		110/220 V
Max output voltage		60/50 Hz
Operating frequency range		100 V
Overall dimensions		25 28,0 kHz
Weight: - generator		520 x 240x 470 mm
- power supply		12,5 kg
Max cable length to connect generator and tool		19 kg
Cooling		80 m
Automatic frequency adjustment		air
Mechanical resonance indication		over entire range
Relative measurement of displacement amplitude of the waveguide output end		light
		needle indicator

### Specifications of Ultrasonic Tools

Parameter	Operating frequency, kHz		
	27*	36*	44*
Design	For manual treatment For automatic treatment		
Rated consumed power, VA	600-1200	300-800	200-500
Excitation voltage, V	60-110		
Bias current,	10-15	6-10	5-8
Oscillating amplitude of the waveguide output end, micron	35-40	30-35	25-30
Treatment speed in manual mode, m/min (m/h)	0,3 1,5 (18 90)		
Treatment speed in automatic and semi-automatic mode, m/h	3 30		
Overall dimensions of the manual tool, mm	455 <sub>1</sub> 85 <sub>1</sub> 80	380 <sub>1</sub> 110 <sub>1</sub> 50	330 <sub>1</sub> 100 <sub>1</sub> 40
Manual tool weight, kg	3,5	1,2	0,75
Cooling	Liquid		
Replaceable tool heads	Straight, angle		
Indenter diameter, mm	2-5		
Hardness of the indenter work face	62-64 HRC		

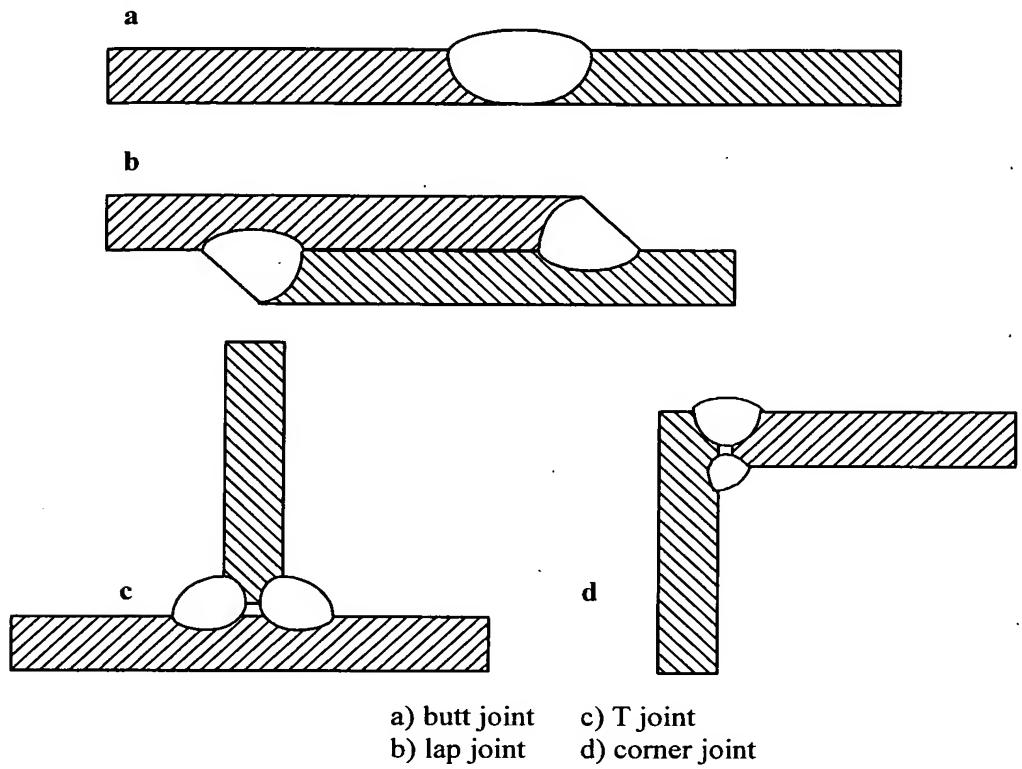
\*Note: 27 kHz production unit ; 36 & 44 kHz prototypes.

Manual ultrasonic tools installed on the welding machine travelers can be usable for automatic UIT. Special tools can be designed based on the standardized ultrasonic transducers developed and manufactured by NSTC (Fig. 3).

## 5. Types of Welded Joints

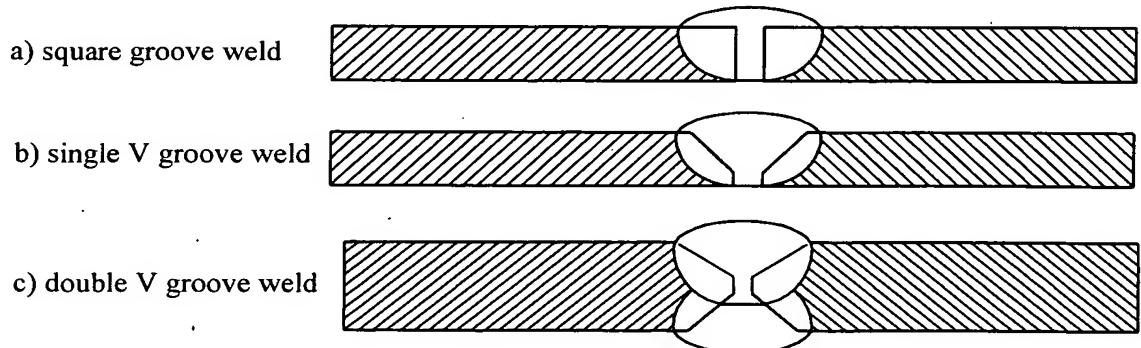
In actual practice, UIT has demonstrated the possibilities of applying this method in production, assembly and repair of essentially all basic welded joint types presented in Fig. 4 8.

5.1 Field and repair welded joints are made in all welding positions: flat, horizontal-vertical, overhead and vertical.



**Fig. 4 Basic Types of Field Welded Joints**

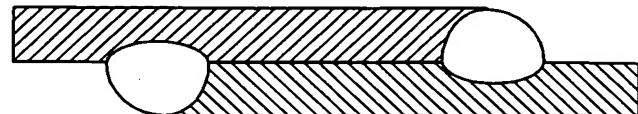
5.2 Butt joints are made both with one and two-sided welds. Square and groove preparation is usable (Fig. 5).



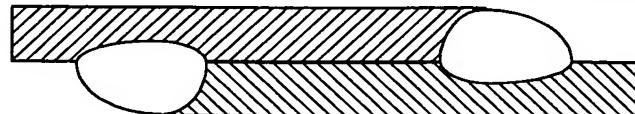
**Fig. 5 Welded butt joints**

5.3 Lap joints have flat faced and convex welds with different leg (overlap) ratio as depicted in Fig. 6. In order to reduce stress concentration, the transversal welds are made with 1:2.5 overlap ratio with subsequent grinding (reinforcement removal).

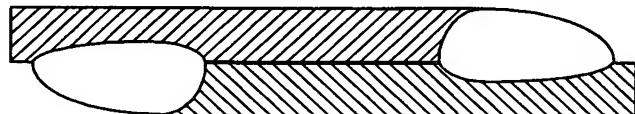
a) 1:1 overlap ratio



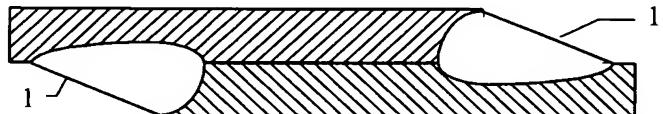
b) 1:1.5 overlap ratio



c) 1:2.5 overlap ratio



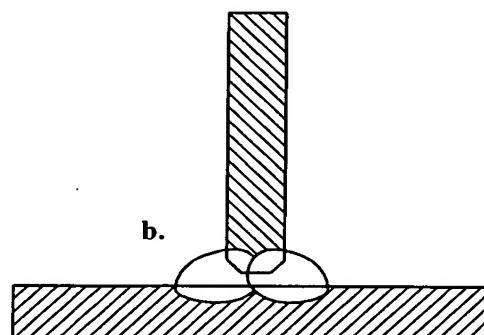
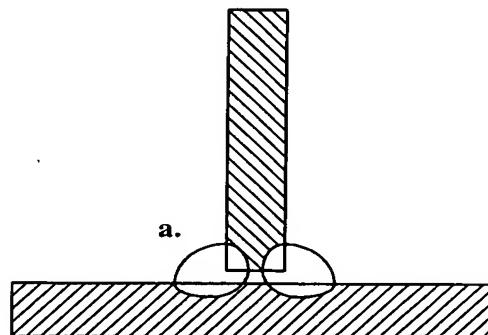
d) 1:2.5 overlap ratio with grinding



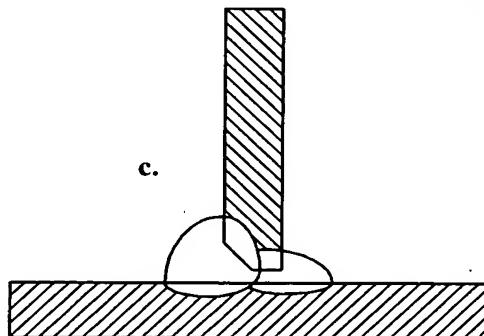
1 ground surface

**Fig. 6 Welded lap joints with different leg ratio**

5.4 Fillet welds in T-joints are made either without groove preparation or as single and double bevel welds (Fig. 7).

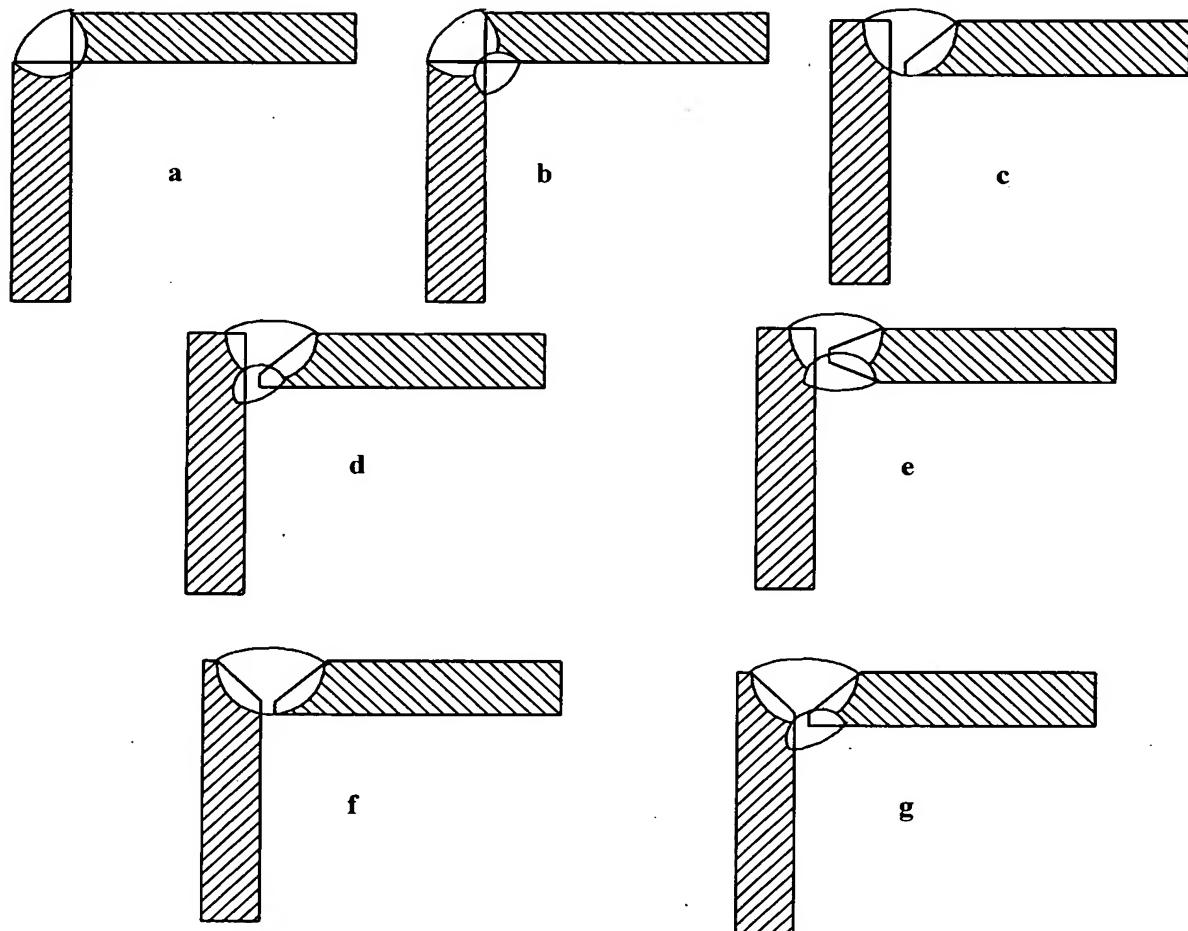


- a) two-sided square weld
- b) two-sided double bevel weld
- c) two-sided single bevel weld



**Fig. 7 Welds in T-joints**

5.5 Welds in corner joints (Fig. 8) may be one and two-sided. Square, single and double bevel preparation is used.



a) outside weld  
b) outside and inside welds  
c, f) single bevel one-sided weld

d, g) single bevel two-sided weld  
e) double bevel weld

**Fig. 8 Welds in Corner Joints**

## 6. Welded Joints Complications

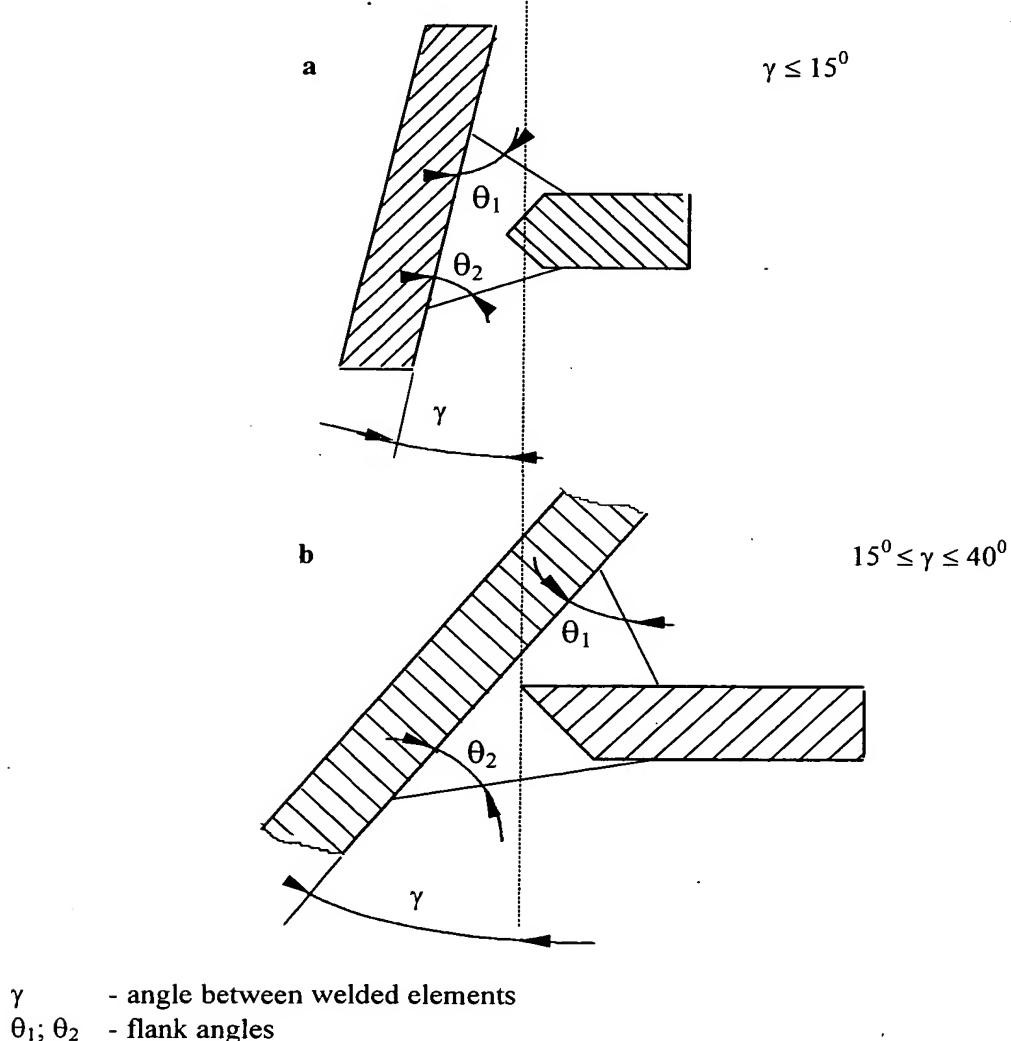
In the development of the UIT application, the designer shall define most loaded and critical welded joints. Thereupon a decision to apply UIT is taken and a list of associated welded assemblies is made out.

The list includes two assembly classes:

- Class 1: assemblies subject to the most unfavorable conditions;
- Class 2: other assemblies.

The set up for UIT application is selected depending on the loading condition, configuration and size of the welded assembly (see paragraph 7.6).

Welds, specifically in T-joints, with flank angle  $\theta$  over  $60^\circ$  have the most unfavorable geometry (Fig. 9) with high service stress concentration factor. In normal practice, such geometry of the weld is optimized by grinding or TIG dressing.



**Fig. 9 Defining Flank Angles When Setting UIT Type**

## 7. UIT Procedures

**7.1 UIT and its parameters for particular assemblies and joints in welded metal structures are set by developer, designer or technologist.**

This is done on the basis of results of experimental tests of UIT efficiency in specific welded joint type, evaluation or prediction of fatigue limit and fatigue strength (life) of these welded joints, and also considering experience in fabrication and maintenance of welded structures with use of UIT.

**7.2 UIT treated welds are indicated by  symbol.**

This symbol is drawn on the reference line as shown in Fig. 10. UIT is applied over the entire length of the marked weld (Fig. 10a) or to the weld section of length defined from start of the weld (Fig. 10b).

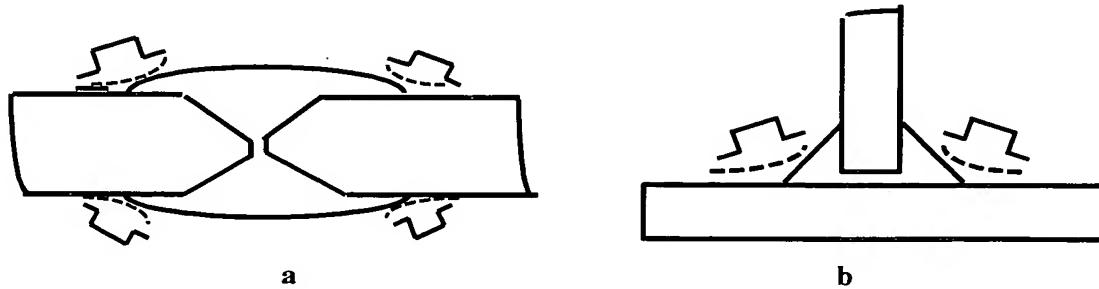
UIT variation for the weld is indicated by symbol drawn on the reference line or in the drawing specification as per Fig. 12.



**Fig. 10 Marking of UIT treated welds**

**7.3 Butt joints are treated by UIT at both sides of the weld toes and from both sides of the welded joint (Fig. 11a).**

Typically, transition between the weld and load-carrying structural member is treated by UIT in fillet welds in corner, lap and T joints. Two-sided welds in T joints are treated from both sides (Fig. 11b).

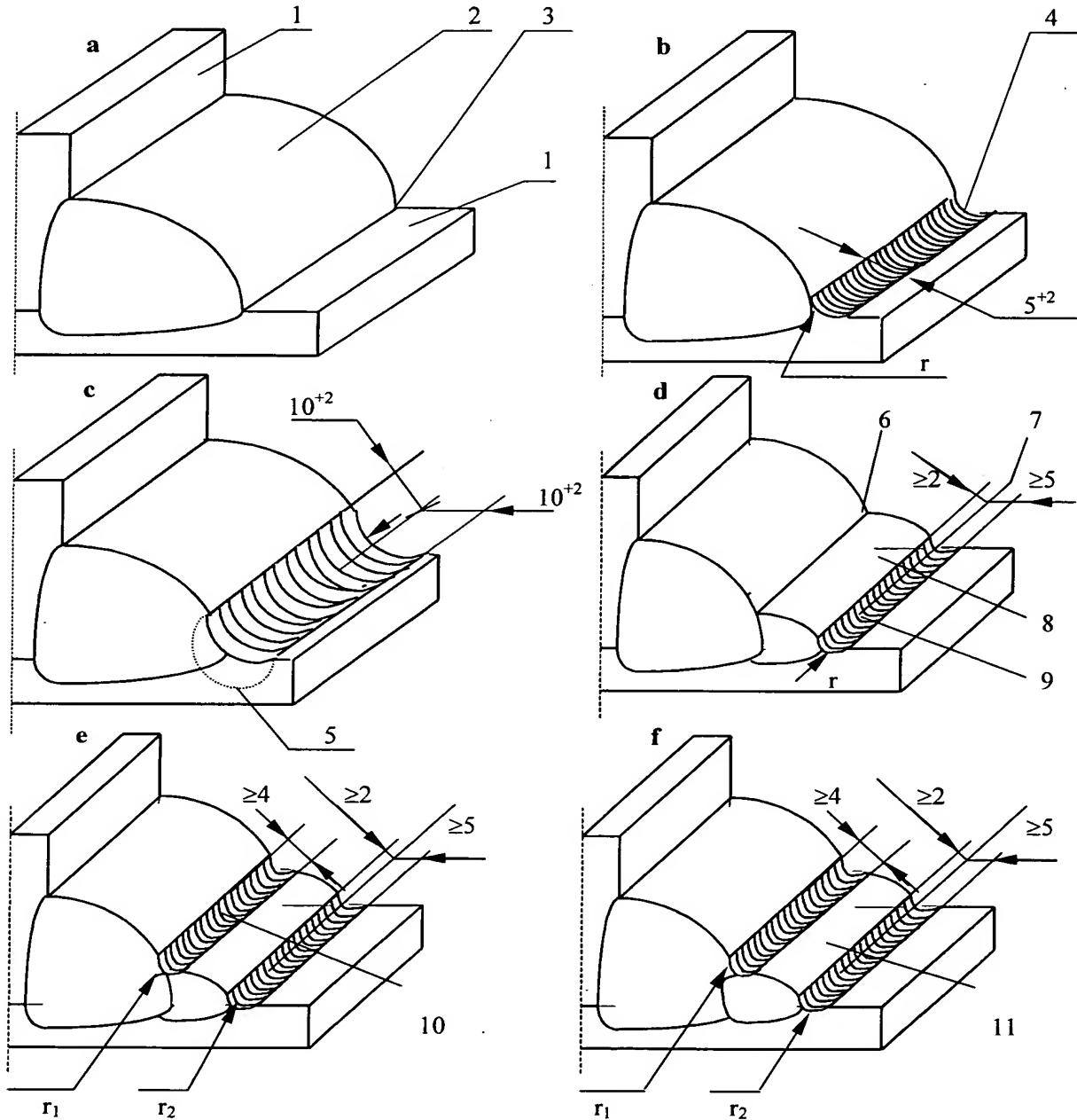


**Fig. 11 UIT Areas in Welds**

**7.4 UIT variation for specific welded assemblies can be provided and described in the drawing specification, procedures or separate sketches.**

Fig. 12 illustrates UIT variations. As an example, T joints are shown.

1) base metal; 2) weld metal; 3) weld toe; 4) UIT treated weld toe; 5) TIG dressing area; 6) fusion line between additional bead and weld metal; 7) fusion line between additional bead and base metal; 8) additional bead (preliminary deposition); 9) UIT treated transition between the additional bead and the base metal; 10) UIT treated transition between the additional bead and the weld metal; 11) additional bead (deposition after making of a weld).



- as welded;
- formation of a smooth transition (groove), 3 - 5mm in radius, at the weld toe by UIT;
- making of a transition of large radius  $r = 5-8\text{mm}$  by TIG dressing and UIT;
- preliminary deposition of the additional bead and UIT of the transition between the additional bead and the weld metal;
- preliminary deposition of the additional bead and UIT at the outer toes of the bead;
- additional bead deposition after making of a weld. UIT at the outer toes of the bead.

**7.5 Effective operating stress deconcentration radius is formed at the weld toe by UIT differs from deconcentration radius formed by any other technique.**

The difference is that the effective value of deconcentration radius after UIT is defined by the geometry of the surface generated by the transfer of residual compressive stresses equal to yield strength into the area of elastic compressive stresses. This radius is at minimum 3 times greater than the deconcentration radius at the UIT treated surface (see Fig. 16).

**7.6 UIT type is selected in accordance with the following Table depending on the assembly class and the flank angle:**

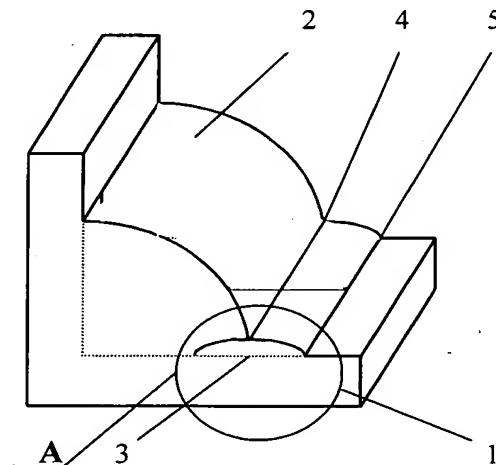
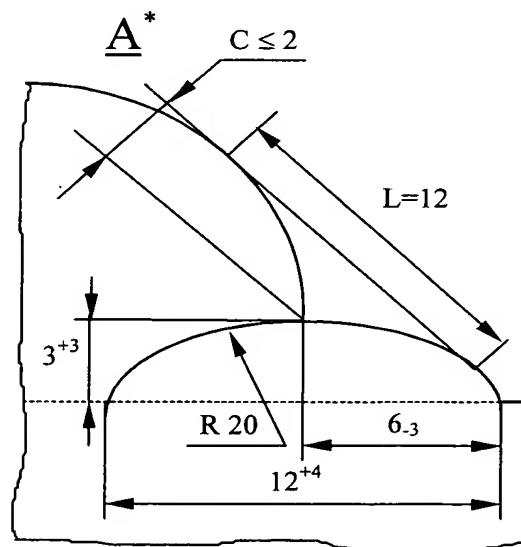
Assembly class as per clause 6	Symbol of UIT variation	
	Flank angle $\theta < 60^\circ$	Flank angle $\theta > 60^\circ$
1	d) and f) variations	e) and f) variations
2	b) variations	e) and f) variations

UIT variation shown in Fig. 12c is auxiliary and best to be used if UIT is applied at pre-melted parts of welds.

**7.7 Additional bead geometry should correspond to Fig. 13.**

Additional bead is made using welding parameters and electrode grade identical to those for making of the weld.

- 1 base metal;
- 2 weld metal;
- 3 additional bead;
- 4 fusion line between the additional bead and the weld metal;
- 5 fusion line between the additional bead and the base metal;

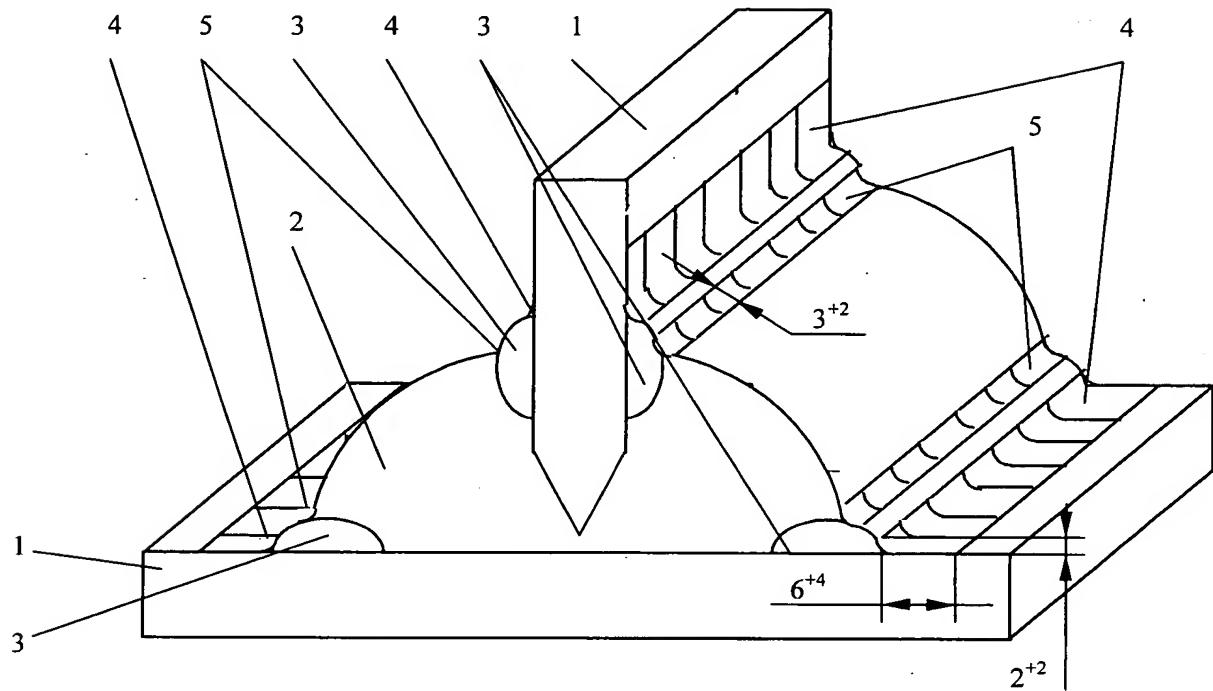


\* Additional bead dimensions are indicated just as an example for UIT application and do not take into account all possible requirements for the structural arrangement of the welded joint.

L gauge length  
depth of bead space

**Fig. 13 Joint form with preliminary deposition of additional bead**

**7.8 UIT of T-joint subjected to extreme conditions is shown in Fig. 14.**



- 1 base metal
- 2 weld metal
- 3 additional bead
- 4 UIT treated transition between the additional bead and the base metal
- 5 UIT treated transition between the additional bead and the weld metal

**Fig. 14 UIT of T joint subjected to unfavorable conditions**

**7.9 UIT parameters for steel are selected in the following ranges:**

excitation frequency	27 kHz
oscillation amplitude	30 - 40 micron
indenter diameter	2-5 mm
treatment speed (tool travel)	18-90 m/h

In the process of treatment the manual ultrasonic tool is located at right angles to the treated surface and pressed against the surface with an axial force of 20-40 N (2-4 kg). This force, as a rule, is produced by tool weight. UIT is used with translational or reciprocal movement of the tool along the weld toe until specified geometry of the treatment area is formed.

Techniques and tooling should ensure access for indenters to the weld toe. With sharp transition in the area of the weld toe, the indenters 1-2 mm in diameter are usable to have access to the weld toe (or indenters 3 mm in diameter with taper sharpening 1 2 mm).

**7.10 Practical requirements for UIT application**

7.10.1 Welding of field joints with UIT application should be performed in accordance with previously designed procedures stipulating the sequence, process, technique and parameters of welding, build-up sequence, sequence and parameters of UIT.

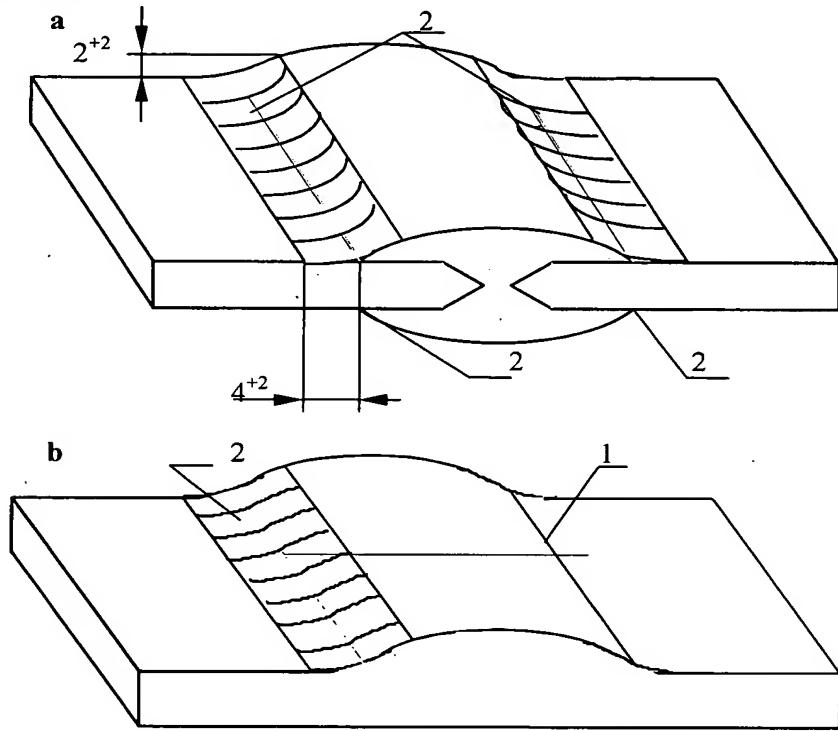
7.10.2 UIT parameters for field welds are set with regard to the joint type, welding position, and welding process. Preliminary deposition of additional beads is added to the process as needed.

7.10.3 With UIT application in the technology of assembly and welding of structures, it is possible to eliminate or reduce preliminary angular change for joints to compensate welding deformations affecting final sizes and shape of the structure. In order to fully compensate

welding deformations UIT parameters are specified in work production plan and checked after welding of initial sections.

7.10.4 Prior to field welding and UIT of structures, as-welded and UIT treated check joints are made to define the mechanical properties of the welded joint, penetration, residual stresses and deformations and assess the efficiency of UIT.

7.10.5 According to operating standards when welding control joints, the base metal, welding consumables, welding and UIT parameters adopted for a given structure should be used. Specimens are tested in accordance with operating standards. Sample joints treated by UIT are also used to make "UIT quality standards (Fig. 15). Strength, plasticity, toughness and hardness of weld metal and heat-affected zone therewith should conform to requirements of operating standards.

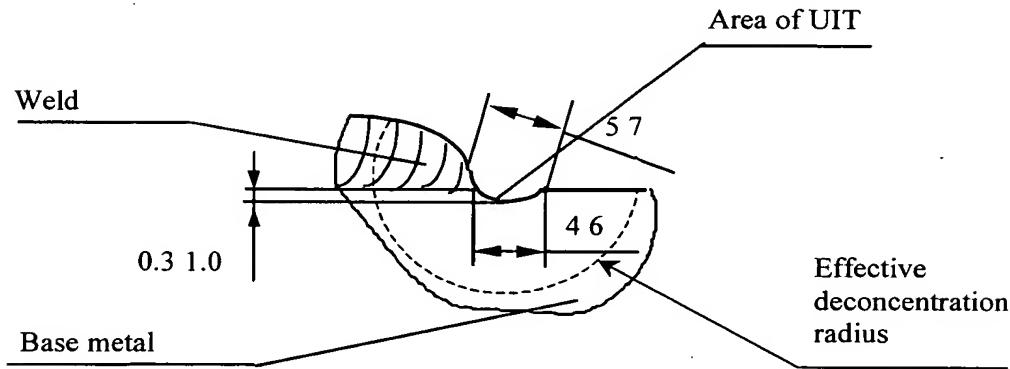


- a) UIT treated butt joint
- b) UIT quality standard for butt joint

- 1 as welded weld toe
- 2 weld toe after UIT

**Fig. 15 UIT of butt joint and UIT quality standard**

7.10.6 The groove is formed at the weld toe during UIT of welded joints (Fig. 16).



**Fig. 16 Recommended groove profile after UIT, specifically for medium strength steel welded joints**

7.10.7 During assembly with UIT of the most critical welded joints, it is recommended that the samples for mechanical tests be made and treated by UIT.

7.10.8 When making multi pass welds, UIT should be applied after each pass. Removing of slag is used after each pass. For this purpose, UIT tool is desirable for use.

7.10.9 Specially trained operators, after study of this guide, as well as design and operating manuals for UIT equipment and tools, are allowed to perform UIT.

7.10.10 UIT quality control is carried out as per requirements stipulated in the design documentation and this document. Surface quality in the UIT area and groove profile should comply with quality standards.

## Conclusion

This Guide is the first version of the document defining the parameters, criteria and variations of UIT application in production and assembly of welded structures. This document may be supplemented and refined with consideration for the results of certification, research and experience of UIT practical applications in actual structures.

## References

1. E.S. Statnikov. Applications of Operational Ultrasonic Impact Treatment (UIT) Technologies in Production of Welded Joints. IIW Doc. XIII-1667-97.
2. Yu.F. Kudryavtsev, V.I. Trufiakov, P.P. Mikheev, E.S. Statnikov. Increasing the Fatigue Strength of Welded Joints in Cyclic Compression. IIW Doc. XIII-1569-94.
3. J.J. Janosch, H. Koneczny, S. Debiez, E.S. Statnikov, V.I. Trufiakov, P.P. Mikheev. Improvement of Fatigue Strength in Welded Joint (in HSS and Aluminium Alloy) by Ultrasonic Hammer Peening. IIW Doc. XIII-1594-95.
4. E.S. Statnikov, L. Kelner J. Baker, H. Croft, V.I. Dvoretski, V.O. Muktepavel. Repair of Fatigue Cracks. IIW Doc. XIII-WG5-18-98.
5. E.S. Statnikov, V.I. Trufiakov, P.P. Mikheev and Yu.F. Kudryavtsev. Specification for weld toe improvement by ultrasonic impact treatment. IIW Doc. XIII-1617-96.
6. V.I. Trufiakov, E.S. Statnikov, P.P. Mikheev and A.Z. Kuzmenko. The efficiency of Ultrasonic Impact Treatment for improving the fatigue strength of welded joints. IIW Doc. XIII-1745-98.
7. P.J. Haagensen, E.S. Statnikov and L. Lopez-Martinez. Introductory fatigue tests on welded joints in high strength steel and aluminium improved by various methods including Ultrasonic Impact Treatment (UIT). IIW Doc. XIII-1748-98.



The International Institute of Welding

# Fatigue design of welded joints and components

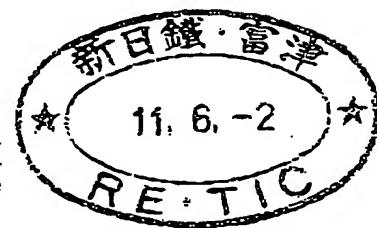
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# 3 FATIGUE RESISTANCE

## 3.1 BASIC PRINCIPLES

Fatigue resistance is usually derived from constant or variable amplitude tests. The fatigue resistance data given here are based on published results from constant amplitude tests. Guidance on the direct use of test data is given in section 3.7 and 4.5.

The fatigue resistance data must be expressed in terms of the same stress as that controlled or determined during the generation of those data.

● In conventional endurance testing, there are different definitions of failure. In general, small specimens are tested to complete rupture, while in large components the observation of a through wall crack is taken as a failure criterion. The fatigue resistance data are based on the number of cycles  $N$  to failure. The data are represented in S-N curves

$$N = \frac{C}{\Delta \sigma^m} \quad \text{or} \quad N = \frac{C}{\Delta \tau^m}$$

In fracture mechanics crack propagation testing, the crack growth rate data are derived from crack propagation monitoring.

All fatigue resistance data are given as characteristic values, which are assumed to have a survival probability of at least 95%, calculated from a mean value of a two-sided 75% confidence level, unless otherwise stated (see 3.7).

## ● 3.2 FATIGUE RESISTANCE OF CLASSIFIED STRUCTURAL DETAILS

The fatigue assessment of classified structural details and welded joints is based on the nominal stress range. The (nominal) stress range should be within the limits of the elastic properties of the material. The range of the design values of the stress range shall not exceed 1.5  $f_y$  for nominal normal stresses or  $1.5 f_y \sqrt{3}$  for nominal shear stresses.

In most cases structural details are assessed on the basis of the maximum principal stress range in the section where potential fatigue cracking is considered. However, guidance is also given for the assessment of shear loaded details, based on the maximum shear stress range. Separate S-N curves are provided for consideration of normal or shear stress ranges, as illustrated in figures (3.2)-1 and (3.2)-2 respectively.

Care must be taken to ensure that the stress used for the fatigue assessment is the same as that given in the tables of the classified structural details. Macrogeometric stress concentrations not covered by the structural detail of the joint itself, e.g. large cutouts in the vicinity of the joint, have to be accounted for by the use of a detailed stress analysis, e.g. finite element analysis, or appropriate stress concentration factors (see 2.2.2).

The fatigue curves are based on representative experimental investigations and thus include the effects of:

- structural stress concentrations due to the detail shown
- local stress concentrations due to the weld geometry
- weld imperfections consistent with normal fabrication standards
- stress direction
- welding residual stresses
- metallurgical conditions
- welding process (fusion welding, unless otherwise stated)
- inspection procedure (NDT), if specified
- postweld treatment, if specified

Furthermore, within the limits imposed by static strength considerations, the fatigue curves of welded joints are independent of the tensile strength of the material.

Each fatigue strength curve is identified by the characteristic fatigue strength of the detail at 2 million cycles. This value is the fatigue class (FAT).

The slope of the fatigue strength curves for details assessed on the basis of normal stresses (fig. (3.2)-1) is  $m=3.00$ . The constant amplitude fatigue limit is  $5 \cdot 10^6$  cycles. The slope of the fatigue strength curves for details assessed on the basis of shear stresses (fig. (3.2)-2) is  $m=5.00$ , but in this case the fatigue limit corresponds to an endurance of  $10^8$  cycles.

The descriptions of the structural details only partially include information about the weld size, shape and quality. The data refer to a standard quality as given in codes and standard welding procedures. For higher or lower qualities, modifications may be necessary as given in 3.5 and 3.8. All butt welds shall be full penetration welds without lack of fusion, unless otherwise stated.

All S-N curves of details are limited by the material S-N curve, which may vary due to different strengths of the materials.

Disregarding major weld defects, fatigue cracks originate from the weld toe, and then propagate through the base material, or from the weld root, and then propagate through the weld throat. For potential toe cracks, the nominal stress in the base

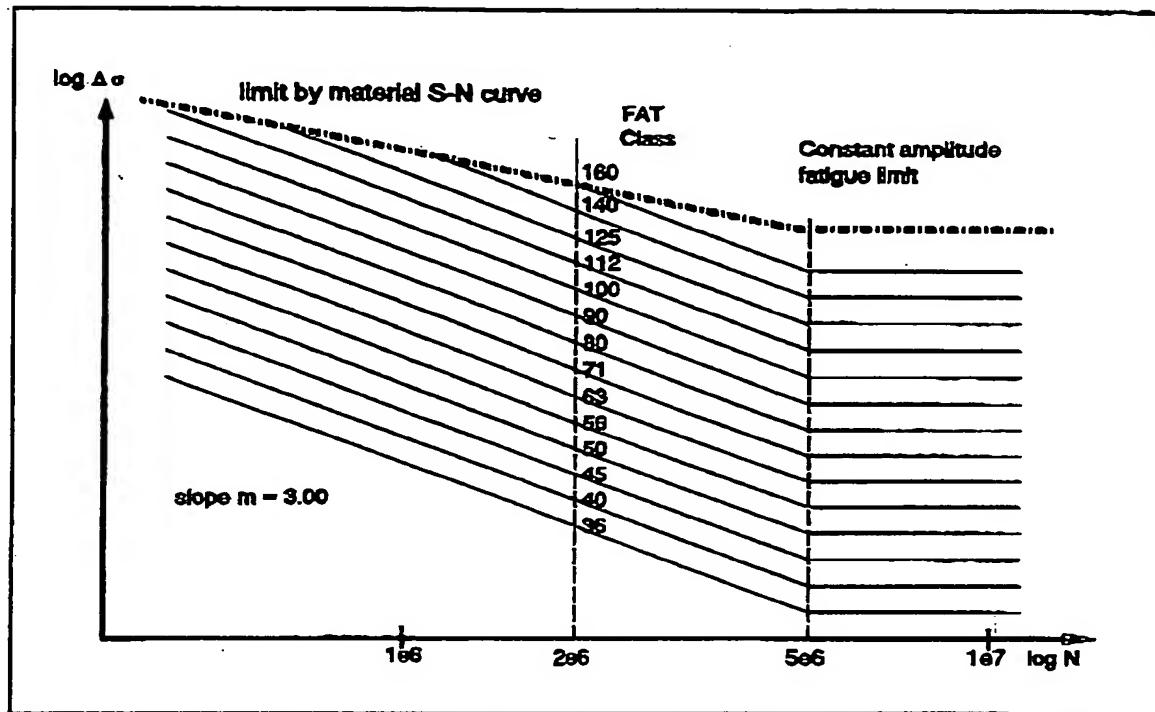


Fig. (3.2)-1: Fatigue resistance S-N curves for  $m=3.00$ , normal stress (steel)

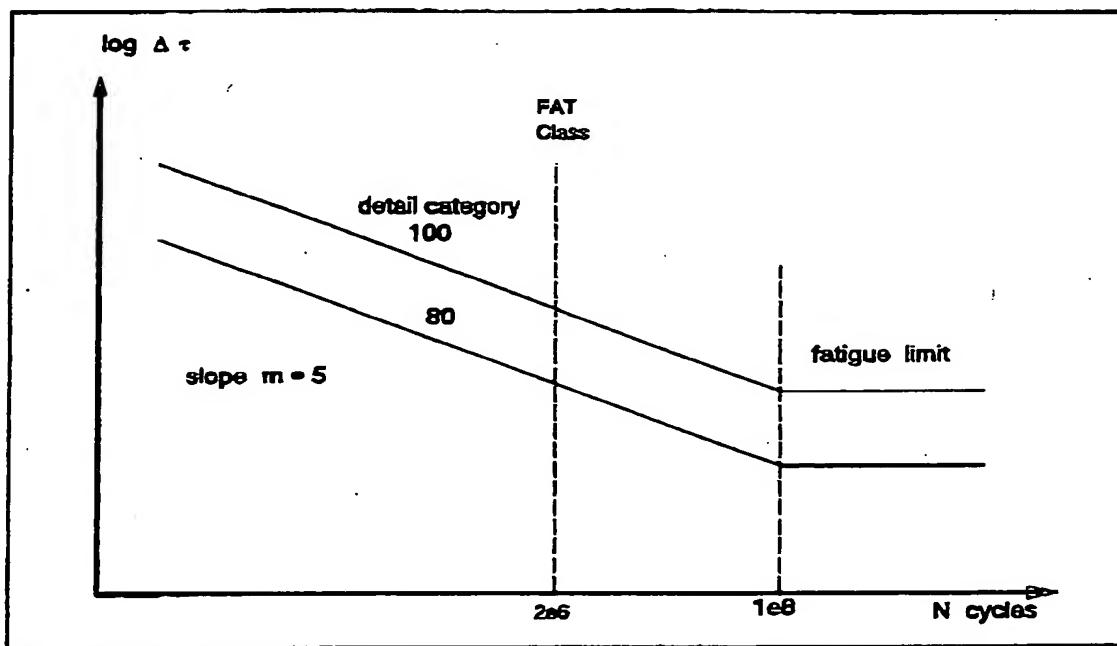


Fig. (3.2)-2 Fatigue resistance S-N curves for shear stress (steel)

material has to be calculated and compared with the fatigue resistance given in the tables. For potential root cracks, the nominal stress in the weld throat has to be

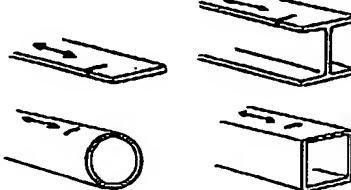
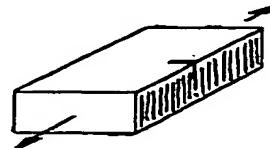
calculated. If both failure modes are possible, e.g. at cruciform joints with fillet welds, both potential failure modes have to be assessed.

### 3.2.1 Steel

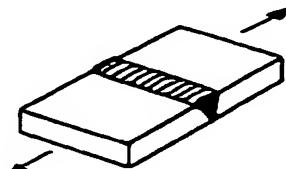
The fatigue resistance values given below refer to welded joints in the as welded condition unless otherwise stated. The effects of welding residual stress and axial misalignment up to  $e/t=0.1$  (see 3.8.2) are also included.

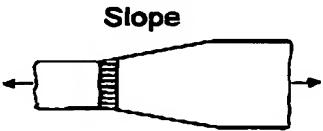
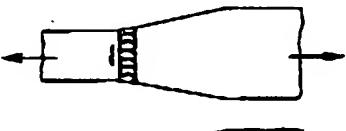
NDT indicates that the weld must be inspected using appropriate methods to ensure that it does not contain any significant imperfections. Arrows indicate the loading direction.

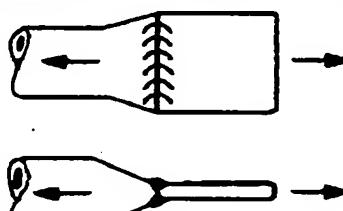
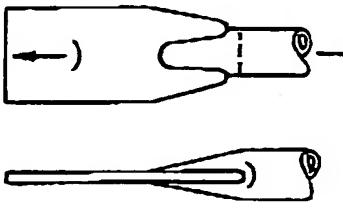
Tab. {3.2}-1: Fatigue resistance values for structural details in steel assessed on the basis of normal stresses.

No.	Structural Detail (Structural steel)	Description	FAT
100	<b>Unwelded parts of a component</b>		
111		<p>Rolled and extruded products</p> <p>1) Plates and flats 2) Rolled sections 3) Seamless hollow sections</p> <p><math>m = 5</math></p> <p>For high strength steels a higher FAT class may be used if verified by test.</p> <p>No fatigue resistance of a detail to be higher at any number of cycles!</p>	160
121		<p>Machine gas cut or sheared material with no drag lines, corners removed, no cracks by inspection, no visible imperfections</p> <p><math>m = 3</math></p>	140

No.	Structural Detail (Structural steel)	Description	FAT
122		Machine thermally cut edges, corners removed, no cracks by inspection  $m = 3$	125
123		Manually thermally cut edges, free from cracks and severe notches  $m = 3$	100
124		Manually thermally cut edges, uncontrolled, no notch deeper than .5 mm  $m = 3$	80
200	<b>Butt welds, transverse loaded</b>		
211		Transverse loaded butt weld (X-groove or V-groove) ground flush to plate, 100% NDT	125
212		Transverse butt weld made in shop in flat position, toe angle $\leq 30^\circ$ , NDT	100

No.	Structural Detail (Structural steel)	Description	FAT
213		Transverse butt weld not satisfying conditions of 212, NDT	80
214		Transverse butt weld, welded on ceramic backing, root crack	80
215		Transverse butt weld on permanent backing bar	71
216		Transverse butt welds welded from one side without backing bar, full penetration.  root controlled by NDT no NDT	71 45

No.	Structural Detail (Structural steel)	Description	FAT
217		Transverse partial penetration butt weld, analysis based on stress in weld throat sectional area, weld overfill not to be taken into account.  The detail is not recommended for fatigue loaded members. It is recommended to verify by fracture mechanics (3.8.5.2)!	45
221		Transverse butt weld ground flush, NDT, with transition in thickness and width  slope 1:5 slope 1:3 slope 1:2	125 100 80
222		Transverse butt weld made in shop, welded in flat position, weld profile controlled, NDT, with transition in thickness and width:  slope 1:5 slope 1:3 slope 1:2	100 90 80
223		Transverse butt weld, NDT, with transition on thickness and width  slope 1:5 slope 1:3 slope 1:2	80 71 63

No.	Structural Detail (Structural steel)	Description	FAT
931		Tube-plate joint, tubes flattened, butt weld (X-groove) Tube diameter < 200 mm and plate thickness < 20 mm	71
932		Tube-plate joint, tube slotted and welded to plate tube diameter < 200 mm and plate thickness < 20 mm tube diameter > 200 mm or plate thickness > 20 mm	63 45

Tab. {3.2}-2: Fatigue resistance values for structural details in steel assessed on the basis of shear stresses.

Structural detail	FAT class	$\log C$ for $m=5$	stress range at fatigue limit [ $N/mm^2$ ]
Parent metal, full penetration butt welds	100	16.301	46
Fillet welds, partial penetration butt welds	80	15.816	36